



Development and Qualification of Space Flight Hardware for Optical Systems

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https://photonics.gsfc.nasa.gov

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Meet the Photonics Group of NASA Goddard Over 20 years of space flight hardware development, testing, & integration





Back row L-R:Erich Frese, Joe Thomes, Marc MatyseckMiddle row L-R:Rick Chuska, Eleanya Onuma, Cameron Parvini, Rob SwitzerFront row L-R:Hali Jakeman, Melanie Ott, Diana Blair,



Trevon Parker





Alexandros Bontzos



Alejandro Rodriguez

All great things require a great team! https://photonics.gsfc.nasa.gov





- History of some of our successful remote and earth missions.
- Here is how we accomplished TRL 9 mission success with commercial off the shelf components (COTS).





Incoming Inspection Integration & Test
Working with our industry partners to provide full service
From incoming inspection of materials all the way to integration of hardware.
Our missions require rigor in our work and documentation.



Custom Spaceflight Optical & Optoelectronic Subsystems using Commercial Components







How Do You Develop and Fabricate Hardware?





<u>Risk mitigation to reduce cost - use space flight component failure mode knowledge;</u> Design out what you can –through configuration; packaging, materials, processes, screening.



Planetary and Earth Orbiting LIDARS Mercury



Mercury Laser Altimeter on Mercury Surface, Space Environment, Geochemistry and Ranging (MESSENGER); development 1999-2003, built by NASA Goddard Space Flight Center Launch 2004, Operation 2011-2015 (travel time 7 years, 4 years usage, decommissioned in 2015)





Planetary and Earth Orbiting LIDARS Mercury



The 24 Million Km Link with the Mercury Laser Altimeter

Jay Steigelman Dave Skillman Barry Coyle John F. Cavanaugh Jan F. McGarry Gregory A. Neumann Xiaoli Sun Thomas W. Zagwodzki Dave Smith Maria Zuber Smith, D. E., *et al.*, Two-way laser link over interplanetary distance, *Science*, 311, 5757, 53, Jan. 2006.

On the way to Mercury a link between NASA GSFC Greenbelt Station and the MLA was established - Longest Laser Link in Space Flight @ 24 Million Km.

MOLA Science Team Meeting Bishop's Lodge, Santa Fe, NM August 24-25, 2005

The success of this experiment led the way for the Laser Ranging investigation on the Lunar Reconnaissance Orbiter.



Planetary and Earth Orbiting LIDARS The Moon https://lunar.gsfc.nasa.gov



Laser Ranging Experiment & Lunar Orbiter Laser Altimeter (LOLA) –Lunar Reconnaissance Orbiter (LRO) Developed 2005-2008; Launch 2009, lifetime requirement 14 months, 3 years desired, actual 8 years and counting.....

LASER RANGING @ 532 nm -Stations Around the World Transmitting to the receiver telescope/ 7 optical fiber array





The assemblies traverse two moving gimbals, and a deployable mandrel 10 meters away to LOLA.



Lunar Orbiter Laser Altimeter (LOLA) Measuring moon topography @ 1064 nm with a 5 fiber array



LRO Fiber Optics LOLA Flight Assembly





Planetary and Earth Orbiting LIDARS

Earth

https://icesat.gsfc.nasa.gov

NASA

Ice, Cloud and Land Elevation Satellite (ICESat-2) - (ATLAS) Advanced Topographic Laser Altimeter System (2012 - 2018) Launched 2018, currently in operation. Expected lifetime > 3 years – measuring the height of sea ice to within an inch.



ATLAS uses ranging measurements with 532 nm and has a sophisticated real time, calibration system.

> 25 simplex, 4 bundle/array to fan out assemblies, ESD compliant-5 different types of fiber; dual and quad fiber arrays; 52 interconnections. **Commercial LED - on board calibration system Fibertek lasers**



Melanie Ott (fiber system lead) inspecting the final flight configuration for fiber optic system. Transmission requirement of >98% for optical fiber receiver system.

Reference: http://icesat.gsfc.nasa.gov



Planetary and Earth Orbiting LIDARS Earth



GEDI: Global Ecosystem Dynamics Investigation LIDAR (2016-2018) Launched Dec 2018, operating currently on International Space Station

#	GEDI Subsystem	Hardware Deliveries
1	Checkout Equipment	Development, fabrication & integration: laser & detector test rack used for qualification of flight instrument, TVAC fiber assemblies down to -120°C.
2	Detector Qualification	Qualification of engineering & flight unit detectors
3	Laser Beam Dithering Unit	Development, fabrication, qualification & integration of engineering and flight units
4	Optical Laser Components	Development, qualification & fabrication of flight laser fiber optic feedthrough. Incoming inspection of laser components.
5	Flight Fiber Optic System	Development, qualification & integration of flight 600/600µm fiber optics transmission >97%; 200/220µm triple fiber arrays for start pulse. Adapter inspections and screening.





Science, Rovers and Communications Mars





Mars Curiosity Rover; ChemCam Instrument Launch Nov. 2011, currently in operation. Hali Jakeman inspects the flight Mars2020 assemblies





Mars 2020 Rover, SuperCam Instrument Currently in integration and test.



Development, fabrication, qualification of flight hardware delivery for JPL

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Communications: Multimode and Singlemode;

- Express Logistics Carrier on International Space Station. Qualification of transceivers, fiber optic assemblies (2006 2010)
- Lunar Laser Communications Demonstration for MIT LL (2010)
- Communications for Cloud Aerosol Transport System; cats.gsfc.nasa.gov (2014) w/ FiberTek, Micropac
- Laser Communications Relay Demonstration; Screening and qualification (laser diodes & photonic components) (2014); Gooch & Housego

Science: Infrared, and/or polarization maintaining, single and multimode, thermal vacuum and cryogenic applications:

- James Webb Space Telescope; Ball Aerospace, Johnson Space Center & GSFC. (2008-2018)



Rob S. @ Ball installs cryo assemblies



Eleanya Onuma installs vacuum feedthroughs



Rob Switzer and Melanie Ott, ELC integration @ Kennedy Space Center



The Future Perspective

https://spacenews.com/is-the-gateway-the-right-way-to-the-moon/







Gateway Roadmap https://spacenews.com/is-the-gateway-the-right-way-to-the-moon/





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COTS Technology Assurance Approach



System Requirements	 Define Critical parameters Define acceptable performance parameters for post test Define components of modules to be tested Define number of samples to test 	
Parts Selection	 Construction Analysis Knowledge of materials Knowledge of construction design, physical analysis Destructive physical analysis (FEA for active parts) 	
Critical Components		
Failure Modes Study	• Components • Modules	
Test Methods	• Capture largest amount of failure modes while testing for space experiment	
Qualification Test Plan(s)	• Contains necessary testing for mission while monitoring for failure modes	

* *Photonic Components for Space Systems*, M. Ott, Presentation for Advanced Microelectronics and Photonics for Satellites Conference, 23 June 2004.



COTS Space Flight "Qualification"





* *Photonic Components for Space Systems*, M. Ott, Presentation for Advanced Microelectronics and Photonics for Satellites Conference, 23 June 2004.

We perform selection, test and qualification of laser components the way the Parts Lab supports EEE parts.







Total Dose Radiation PerformanceNot usually a detriment but for calibration purposes is always necessary;
Example Mercury Laser Altimeter Optical Fiber Radiation Data





Flexlite Radiation Test, 22.7 rads/min at –18.3°C

Flexlite Radiation Test, 11.2 rads/min at –24.1°C

Radiation Conclusion: < .07 dB, using 11.2 rads/min, -24.1°C, 26.1 in, "dark" Results for 10 m, at 30 Krads, -20°C, 850 nm, 23 rads/min ~ 1 dB or 0.10 dB/m



Thank You to Our Partners! (not all are listed here)





And thank you for your time.